

Pseudo Orbit Data Assimilation (PDA)

PDA cost function

Perfect & Imperfect Model Scenario

Error growth and dynamical Consistency

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$$\frac{d C(\mathbf{U})}{d \mathbf{u}_{t}} = 2 \cdot \begin{cases} (u_{t} - F(u_{t-1})) + (u_{t+1} - F(u_{t})) d_{t}F(u_{t}) & t = n+1 \\ -(u_{t} - F(u_{t-1})) + (u_{t+1} - F(u_{t})) d_{t}F(u_{t}) & -n+1 < t < 0 \\ -(u_{t} - F(u_{t-1})) & t = 0 \end{cases}$$

approaches to a trajectory of the model asymptotically









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 \widetilde{x} true state $x = g(\tilde{x})$ **Projection into model state space** projected true $x \in \Re^m$ state **Observation** $\mathbf{s}_{t} = h(g(\mathbf{\widetilde{x}}_{t})) + \mathbf{\eta}_{t}$ observation $h(\cdot)$ operator $\mathbf{\kappa}_t = \mathbf{s}_t - h(\mathbf{u}_t)$ implied noise Point-wise Model Error Imperfection Error $\mathbf{e}_{t} = g(\mathbf{\widetilde{x}}_{t+1}) - F(g(\mathbf{\widetilde{x}}_{t}))$ $\boldsymbol{\omega}_t = \mathbf{u}_{t+1} - F(\mathbf{u}_t)$

PDA stopping criterion

It is desirable to obtain **pseudo-orbits** whose **implied noise** and **imperfection error** are consistent with the **observational noise** and the **point-wise model error** respectively.







(c) is the RMS distance between the pseudoorbit and the target pseudo-orbit.









Imperfection Error (a) after 5 GD iterations, (b) after 40 GD iterations, (c) after 200 GD iterations. The color reflects the difference between the imperfection error and the corresponding pointwise model error. (d) The evolution of the slope of the best fit line relating the imperfection error and the corresponding point-wise model error







Observational noise & model inadequacy

- PDA stopped when the standard deviation of the implied noise first exceeded the standard deviation of the observational noise.
- When the observational noise is much smaller than the point-wise model error, the latter can be well estimated by the imperfection error.
- When the observational noise is significantly bigger than the point-wise model error, the imperfection error appears more random

"It is stressed that state dependent model error information is an output of the proposed approach, whereas both EnKF and WC4DVAR require specifications and/or assumptions as an input."







SV – red, BV – blue, Random Field Pert. – Green, Random Pert. - black



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PDA cost function leads to trajectories in the model state space

PDA can be used to provide more dynamically consistent initial states

Use it as a postprocessing tool for EnKF - Pseudo Orbits







The following slights are optional



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